



SUPPLEMENTAL TYPE CERTIFICATE

10036114, REV. 1

This Supplemental Type Certificate is issued by EASA, acting in accordance with Regulation (EC) No. 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to

AIRWOLF AEROSPACE L.L.C.

**15369 MADISON ROAD
MIDDLEFIELD OH 44062
USA**

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and environmental protection requirements when operated within the conditions and limitations specified below:

Original Type Certificate Number : EASA.IM.R.512
Type Certificate Holder : BELL HELICOPTER TEXTRON CANADA
Type Design - Model : 206A, 206B, 206L
Type Design - Model : 206L-1, 206L-3, 206L-4
Original STC Number : FAA STC SR02964CH

Description of Design Change:

TENSION-TORSION STRAP - REPLACEMENT OF LIFE-LIMITED PART
Amended Airworthiness Limitation Section of the Instruction for Continued Airworthiness AA-206-ICA

EASA Certification Basis:

The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.

See Continuation Sheet(s)

For the European Aviation Safety Agency,

Date of issue: 28 February 2013


Massimo MAZZOLETTI
Certification Manager
Rotorcraft, Balloons, Airships

Note:

The following numbers are listed on the certificate:
EASA current Project Number: 0010011401-001

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Associated Technical Documentation:

Airworthiness Limitation Section of the Instruction for Continued Airworthiness AA-206-ICA Rev. E, FAA Approved on 19 November 2012;
or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) and/ or the Technical Implementation Procedures of EU/ USA Bilateral Agreement.

Bell 206A and 206B:

Install in accordance with Bell Helicopter Manual BHT-206A/B-M&O-1 - Maintenance and Overhaul Instructions revision 41 dated 21 November 1995 subparagraph (i)(7) on page 2-16;

Bell 206L, 206L-1, 206L-3, and 206L-4:

Install in accordance with Bell Helicopter Manual BHT-206L-CR&O-3 - Component Repair and Overhaul Manual Volume 3 Main Rotor/Main Rotor Drive system revision 3 dated 10 January 2010 paragraph 62-35 (Assembly), subparagraph 5 and 6.

All models:

Instructions for Continued Airworthiness for Airwolf Replacement PMA Main Rotor Blade Tension-Torsion Strap Assemblies on BHT 206 & OH-58 Model Series Rotorcraft AA-206-ICA revision E dated 17 October 2012.

Limitations/Conditions:

Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

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Note:

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